UDC 629.7.066.3

Siberian Journal of Science and Technology. 2018, Vol. 19, No. 2, P. 246–250

## DEVELOPMENT OF THREE-POINT AVIATION FUEL QUANTITY GAUGE

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In response to the well-developed digital technologies, modern methods can be worked out for the spheres where radical changes seemed hardly possible. This paper describes the development of a new method of fuel quantity measurement that has not been applied before. To measure the level of fuel in the fuel tank of an aircraft, it is proposed to use three fuel level gauges and a special electronic calculation unit; they model the fuel level surface inside the given volume, and then the actual amount of fuel can also be calculated. This can considerably reduce the evaluation errors allowed with the application of the existing fuel quantity gauges. The main advantage of the system offered is the elimination of the errors arising with the aircraft evolutions and irregular motions.

The article gives the analysis of the fuel level assessment methods used in the aviation sphere at present, the types of fuel quantity gauges used in aviation, and the specific conditions of measuring the fuel level in the aircraft fuel tanks.

The proposed method has a number of advantages, in comparison with the traditional ways of measuring the fuel level; a basic mathematical model of the aircraft tank fuel level calculation has also been worked out.

Keywords: fuel quantity gauge, fuel tank, measurement, error, aircraft.

Сибирский журнал науки и технологий. 2018. Т. 19, № 2. С. 246-250

## РАЗРАБОТКА ТРЁХТОЧЕЧНОГО АВИАЦИОННОГО ТОПЛИВОМЕРА

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Благодаря хорошо развитым на данный момент цифровым технологиям становится возможным создание новых, современных методов в тех сферах, где, казалось бы, уже больше нечего менять. Данная статья посвящена разработке нового, несуществующего на данный момент, метода измерения топлива. Для измерения топлива предлагается использовать три топливных датчика и вычислительный элемент для моделирования положения уровня топлива в пространстве с дальнейшим расчетом объёма топлива, что позволит уменьшить погрешности, возникающие при эксплуатации топливомеров. Главным преимуществом данной системы будет устранение погрешности, возникающей при эволюциях воздушного судна, а также при его неравномерном движении.

Рассмотрены способы замера уровня топлива, используемые в авиационной сфере в настоящее время, виды топливомеров, применяемых в авиации, а также особенности измерения уровня топлива.

Предложенный способ обладает рядом преимуществ, в сравнении с традиционными способами замера уровня топлива, в статье представлена математическая модель, на основе которой будет производиться измерение уровня топлива в топливном баке воздушного судна.

Ключевые слова: топливомер, топливный бак, погрешность, авиация.

Doi: 10.31772/2587-6066-2018-19-2-246-250

**Introduction.** Fuel quantity gauge is an instrument that measures the volume or weight of fuel or oil in tanks. It is used to measure the level of fuel in land transport and aircraft, in contrast to level gauges measuring the level of liquids or bulk materials in various tanks and storage facilities.

As a rule, direct measurement of the amount of fuel meets a lot of impediments, that is why indirect measurements are widely used, such as measuring the surface level or fuel pressure inside the tank [1–6].

Among the features of measuring the fuel level, we can note the following:

- fuel tank volumes are limited - that imposes restrictions on the level sensors in use;

 fuel tanks come in different shapes, sometimes quite irregular – that is the reason for using profiled level sensors;

external forces and accelerations cause redistribution of fuel inside the tank – that provokes measurement errors;

- for aircraft with several fuel tanks, it is necessary to relocate the center of fuel load by pumping fuel from one tank into another; that is necessary to stabilize the center of mass of the aircraft [7; 8].

Aviation fuel level gauges. At the present time there are several types of fuel gauges in use, they are divided according to the method of measurement and the field of use. There are two main types of fuel gauges most widely used in aviation:

- capacitive fuel gauges – the operational principle of that gauge type is based on the significant difference in the dielectric properties of air and fuel (fig. 1);

- float-type fuel gauges - the operational principle of that gauge type is based on registering the position of the slider on the variable resistance which is moved by a lever with a float - a plastic or metal hollow unit floating on the surface of the fuel.

Indirect measurements of the fuel level in the tank and conversion of the obtained value into an analog signal make it possible to measure the fuel level in the aircraft tank [9; 10].

Modern fuel gauges have the errors induced by irregular movement of the aircraft and its deviations from the horizontal plane. These errors can be eliminated by applying a different method of fuel measurements.

**Three-point method of fuel measurement.** The aim of this method is elimination of the errors in situation when the fuel surface deflects from its normal position (the level when the fuel is not affected by external forces).

The three-point method of fuel measurement implies the use of three fuel gauges and an electronic calculation unit with pre-specified geometry characteristics of the fuel tank. Thanks to modern digital technologies and thoroughly worked-out mathematical methods, we can make a mathematical and virtual model of the fuel tank [11–14]. Using three fuel gauges we can measure the height of the fuel level in three different locations inside the tank; that allows to determine three points of the surface level of the fuel (fig. 2).

According to the space coordinates of the three points inside the fuel tank, a simulation of the fuel surface plane can be produced by the digital unit. This model plane will cut off the upper (empty) part of the fuel tank. With the help of the specified mathematical methods, the form of the modelled truncated figure (truncated fuel tank) allows to calculate its volume. To make the calculation, it is necessary to divide the fuel tank into identical segments and state the fuel level value in each segment. According to the obtained parameters of the area of the segment and the fuel level in it, the volume of the fuel content in this segment can be determined (fig. 3).

Mathematical model of a three-point fuel level gauge. To determine the fuel level height in each segment, we use equation of plane for three points [15]:

$$\begin{aligned} (x-x_0) \begin{vmatrix} y_1 + y_0 & y_2 - y_0 \\ z_1 + z_0 & z_2 - z_0 \end{vmatrix} - (y-y_0) \begin{vmatrix} x_1 + x_0 & x_2 - x_0 \\ z_1 + z_0 & z_2 - z_0 \end{vmatrix} + \\ &+ (z-z_0) \begin{vmatrix} x_1 + x_0 & x_2 - x_0 \\ y_1 + y_0 & y_2 - y_0 \end{vmatrix} = 0; \\ &xA - yB + zC + D = 0; \\ &xA - yB + D = -zC; \\ &z = \frac{-xA + yB - D}{C}, \end{aligned}$$

where A, B and C are the values of the determinants of equation of plane, and D is determined as follows:

$$D = -x_0A + y_0B - z_0C,$$
$$z_{ik} = \frac{-x_iA + y_kB - D}{C}.$$

When the fuel level in each segment is stated, it becomes possible to determine the total volume occupied by the fuel in the fuel tank:

$$V = s \times \sum_{k=1}^{n} \sum_{i=1}^{m} z_{ik} = s \times \sum_{k=1}^{n} \sum_{i=1}^{m} \frac{-x_i A + y_k B - D}{C}$$

*s* is the area of a separate segment, *n*, *m* are the maximum number of segments formed along the *X* and *Y* axes.

However, the calculation of the real model will not coincide with the actual summing up, as the tank is of a confined volume at the top and at the bottom.



Fig. 1. Capacitive fuel gauges Рис. 1. Емкостные топливомеры



Fig. 2. Three-point fuel-content indication method scheme

Рис. 2. Функциональная схема трехточечного топливомера



Fig. 3. Section of the fuel tank volume by the fuel level plane

Рис. 3. Сечение топливного бака плоскостью уровня топлива

The resulting fuel level plane may go across the top and bottom surfaces of the tank, which will cause an error in the calculations.

When calculating each segment, its maximum and minimum values on the Z axis will be taken into account.

Fig. 4 shows the fuel tank horizontal cross-section. The color of the segments corresponds to the fuel level in each segment:

$$-$$
 black  $-$  H = 0;  
 $-$  vellow  $-$  H = Z:

- white - H = max.



Fig. 4. Fuel tank horizontal cross-section

Рис. 4. Топливный бак в горизонтальном сечении

When the fuel level surface deflects from the horizontal plane, some of the tank segments may become empty (black color); then, according to equation of the plane, the value of the z coordinate will be negative, and the value of the volume will also become negative in this case. To avoid this, in case when Z takes a negative value, the given segment will be equated to zero. When the fuel level in the segments comes to its maximum (white color), and according to equation of plane, the z coordinate takes a value greater than maximum, Z will be equated to the maximum value for the given segment.

**Conclusion**. With the given corrections, it becomes possible to make calculations of the greatest accuracy. It is obvious that the virtual truncated model will correspond to the actual level of the fuel in the tank, and the error in case of the external forces effect will not arise, as the virtual model will show the slope of the cross-section plane with no change in the volume.

## References

1. Grigorovskiy B. K., Katsyuba O. A., Priputnikov A. P. [Display a variety of information and measurement process by a model number of fuel engineers. Representativeness of the display]. *Vestnik SAMGUPS*. 2015, Vol. 2, No. 2, P. 150–155 (In Russ.).

2. Dzhezhora A. A., Rubanik V. V., Savchuk V. K., Kuz'minich A. V. [Capacitive level sensors for electroconductive liquid]. *Datchiki i sistemy*. 2008, No. 12, P. 26–29 (In Russ.). 3. Mastepanenko M. A., Vorotnikov I. N., Anikuev S. V. [Mathematical models and methods of processing measuring signals of capacitive DC converters]. *Stavropol', AGRUS.* 2015, 232 p. (In Russ.).

4. Dzhezhora A. A., Rubanik V. V., Savchuk V. K. *Kontrol' urovnya topliva* [Fuel level control]. *Vestnik Polotskogo gosudarstvennogo universiteta*. 2009, No. 2, P. 21–25 (In Russ.).

5. Bogoyavlenskiy A. A. [Instrumental control of the stock and flow of working fluids during technical operation of aircraft]. *Mir izmereniy*. 2017, No. 4, P. 16–23 (In Russ.).

6. Rechkin A. G., Kraynikov V. A., Sablin A. S. [To the question of measuring the fuel stock on board an aircraft]. *II shkola-seminar molodykh uchenykh "Fundamental'nye problemy sistemnoy bezopasnosti"* [II school-seminar of young scientists "Fundamental problems of system security"]. Yelets, 2015, P. 196–202 (In Russ.).

7. Danilov V. G., Shemsedinov I. Sh. [Training system for solving problems in analytical geometry with generating tasks based on generating grammars]. *Kachestvo. Innovatsii. Obrazovanie.* 2009, No. 47, P. 5–10 (In Russ.).

8. Kenmoku Masakatsu. Analytic solutions of the wheeler-dewitt equation in spherically symmetric geometry. *Gravity and cosmology*. 2009, Vol. 5, No. 4, P. 289–296.

9. Pylilo I. S., Klybik V. K. [Selecting a prospective sensor type for continuous fuel level measurement]. *Mekhanizatsiya i elektrifikatsiya sel'skogo khozyaystva*. 2012, No. 4, P. 160–166 (In Russ.).

10. Goncharov D. S., Dzhezhora A. A. [Influence of coaxiality of cylindrical shells of circular section on the capacity of fuel level sensors]. 50 mezhdunarodnaya nauchno-tekhnicheskaya konferentsiya prepodavateley i studentov, posvyashchennaya godu nauki [50th International Scientific and Technical Conference of Teachers and Students, dedicated to the Year of Science]. Vitebsk, 2017, P. 105–106 (In Russ.).

11. Koshevoy N. D., Matveev A. G. [Development of algorithms for modeling the operation of fuel consumption sensors and interaction with the fuel main]. *Radioelektronika, informatika, upravlenie.* 2011, No. 2, P. 54–59 (In Russ.).

12. Vershinin O. S., Sharov V. V. [Experimental method for estimating the error of an automobile fuel level sensor]. *Izvestiya vysshikh uchebnykh zavedeniy. Problemy energetiki.* 2008, Vol. 2, No. 3, P. 116–121 (In Russ.).

13. Gurtovtsev A. L. [About Metrology of Electronic Power Meters]. *Elektro. Elektrotekhnika, elektroenergetika, elektrotekhnicheskaya promyshlennost'.* 2008, No. 2, P. 44–52 (In Russ.).

14. Buzhinskiy V. A. [About fluid oscillations in fuel tanks with damping gratings]. *Kosmonavtika i raketostroenie.* 2007, No. 46, P. 110–120 (In Russ.).

15. Pisarev N. S., Statsenko N. I. [Three-point aviation fuel gauge]. *Materialy XXI Mezhdunar. nauch. konf. "Reshetnevskie chteniya"* [Materials XXI Intern. Scientific. Conf. "Reshetnev reading"]. Krasnoyarsk, 2017, Vol. 1, P. 468–469 (In Russ.).

## Библиографические ссылки

1. Григоровский Б. К., Кацюба О. А., Припутников А. П. Отображение модельным рядом топливомеров многообразия информационно-измерительного процесса. Репрезентативность отображения // Вестник САМГУПС. 2015. Вып. 2, № 2. С. 150–155.

2. Емкостные датчики уровня электропроводящей жидкости / А. А. Джежора [и др.] // Датчики и системы. 2008. № 12. С. 26–29.

3. Математические модели и методы обработки измерительных сигналов емкостных преобразователей на постоянном токе / М. А. Мастепаненко, И. Н. Воротников, С. В. Аникуев. Ставрополь : АГ-РУС, 2015. 232 с.

4. Джежора А. А., Рубаник В. В., Савчук В. К. Контроль уровня топлива // Вестник Полоцкого государственного университета. 2009. № 2. С. 21–25.

5. Богоявленский А. А. Инструментальный контроль запаса и расхода рабочих жидкостей при технической эксплуатации воздушных судов // Мир измерений. 2017. № 4. С. 16–23.

6. Речкин А. Г., Крайников В. А., Саблин А. С. К вопросу измерения запаса топлива на борту воздушного судна // Фундаментальные проблемы системной безопасности : II школа-семинар молодых ученых. Елец, 2015. Р. 196–202.

7. Данилов В. Г., Шемсединов И. Ш. Обучающая система для решения задач по аналитической геометрии с генерацией заданий на основе порождающих грамматик // Качество. Инновации. Образование. 2009. № 47. С. 5–10.

8. Kenmoku Masakatsu. Analytic solutions of the wheeler-dewitt equation in spherically symmetric geometry // Gravity and cosmology. 2009. Vol. 5,  $N_{\rm P}$  4. C. 289–296 (In Eng.).

9. Пылило И. С., Клыбик В. К. Выбор перспективного типа датчика для непрерывного измерения уровня топлива // Механизация и электрификация сельского хозяйства. 2012. № 4. С. 160–166.

10. Гончаров Д. С., Джежора А. А. Влияние соосности цилиндрических оболочек кругового сечения на емкость датчиков уровня топлива // 50-я Междунар. науч.-техн. конф. преподавателей и студентов, посвящённая году науки. Витебск, 2017. С. 105–106.

11. Кошевой Н. Д., Матвеев А. Г. Разработка алгоритмов моделирования работы датчиков расхода топлива и взаимодействия с топливной магистралью // Радиоэлектроника, информатика, управление. 2011. № 2. С. 54–59.

12. Вершинин О. С., Шаров В. В. Экспериментальный метод оценки погрешности автомобильного датчика уровня топлива // Известия высших учебных заведений. Проблемы энергетики. 2008. Vol. 2, № 3. С. 116–121.

13. Гуртовцев А. Л. О метрологии электронных электросчетчиков // Электротехника, электроэнергетика, электротехническая промышленность. 2008. № 2. С. 44–52.

14. Бужинский В. А. О колебаниях жидкости в топливных баках с демпфирующими решетками // Космонавтика и ракетостроение. 2007. № 46. С. 110–120.

15. Писарев Н. С., Стаценко Н. И. Трехточечный авиационный топливомер // Решетневские чтения : материалы XXI Междунар. науч. конф. Красноярск, 2017. Vol. 1. С. 468–469.

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